CRITICAL ITEMS LIST (CIL)

SYSTEM:

ASI

SUBSYSTEM: REV & DATE: ET Interface Hardware J, 12-19-97

FUNCTIONAL CRIT: PHASE(S): HAZARD REF:

5.11

DCN & DATE: ANALYSTS:

C. Rush/E. Howell

FAILURE MODE:

Structural Failure

FAILURE EFFECT:

Loss of mission and vehicle/crew due to collapse of interface system resulting in

fire/explosion or debris source to orbiter.

TIME TO EFFECT:

Immediate

FAILURE CAUSE(S):

Improper Manufacture Failure of Attaching Hardware

REDUNDANCY SCREENS:

Not Applicable

FUNCTIONAL DESCRIPTION: Interface and structural load path between ET and bipod strut.

FMEA ITEM	PART NO.	PART NAME	<u>ατγ</u>	EFFECTIVITY
4.5.43.1	80911009190-130 -140 -500	Strut Assembly, Nonadjustable	1 1 1	LWT-54 thru 83 LWT-84 thru 88 LWT-89 % Up

REMARKS:		
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CRITICAL ITEMS LIST (CIL) CONTINUATION SHEET

SYSTEM: SUBSYSTEM:

FMEA ITEM CODE(S):

ASI

ET Interface Hardware

4.5.43.1

REV & DATE: DCN & DATE: J, 12-19-97

RATIONALE FOR RETENTION

DESIGN:

A, B: The strut end is machined from a TI-6AL-4V ANN titanium alloy casting. Materials are selected in accordance with MMC-ET-SE16 which assures repetitive conformance of composition and properties. Surface integrity is assured by penetrant inspection per STP2501. The strut assembly and attachment hardware are designed to the required ultimate safety factor of 1.4 (ET Stress Report 826-2188).

B: Attaching hardware is selected from the Approved Standard Parts List (ASPL 826-3500), installed per STP2014 and torqued using values specified on Engineering drawings. Tensile installation loads are sufficient to provide screening for major flaws in individual fasteners.

TEST:

The Strut Assembly, Nonadjustable is certified. Reference HCS MMC-ET-TMO8-L-S179 (LWT-54 thru 88) and HCS MMC-ET-TMO8-L-S508 (LWT-89 & Up).

<u>Vendor:</u>

B: Attaching fasteners are procured and tested to standard drawings 26L2 and 33L2.

INSPECTION:

<u>Vendor Inspection - Lockheed Martin Surveillance:</u>

- A, B: Verify materials selection and verification controls (MMC-ET-SE16, drawing 80911009121 and standard drawings 26L2 and 33L2; STM5632 for LWT-54 thru 83; STM5633 for LWT-84 & Up).
- A: Inspect dimensional conformance (drawing 80911009120).
- A: Penetrant inspect part (drawing 80911009120 and STP2501, Type 1, Method A).

MAF Quality Inspection:

- 8: Inspect that attaching hardware is free from damage (drawing 80911009190 and STP2014).
- A, B: Verify fastemer installation and witness torque (drawing 80911009190).

FAILURE HISTORY:

Current data on test failures, unexplained anomalies and other failures experienced during ground processing activity can be found in the PRACA data base.